

Have a Safe Flight: Bon Voyage!



Mariela Buchin, Wonron Cho, Scott Fisher

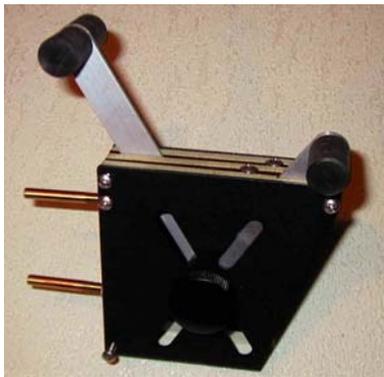
# Making the “Smart Flight Vest”

- Mount two angular rate sensors onto the upper body of the flight vest
- Separate device will measure throttle

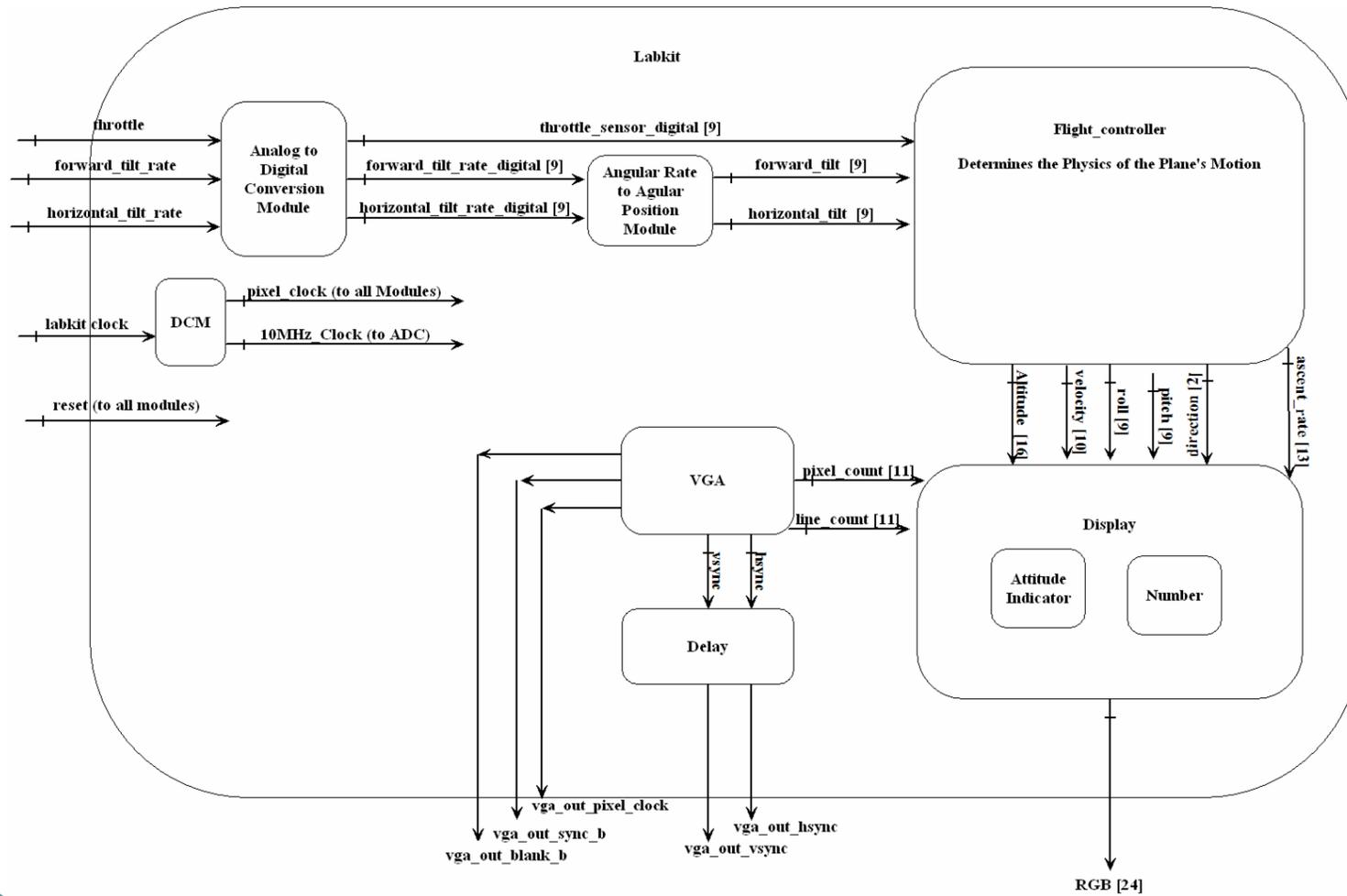


# Controlling Throttle

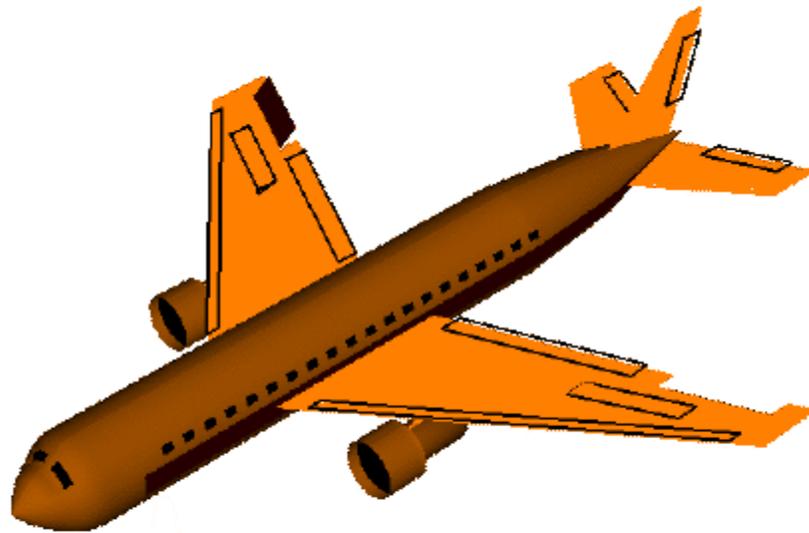
- Want functionality of being able to adjust and set throttle
- Will mount a handle onto resistor arm to imitate a throttle lever



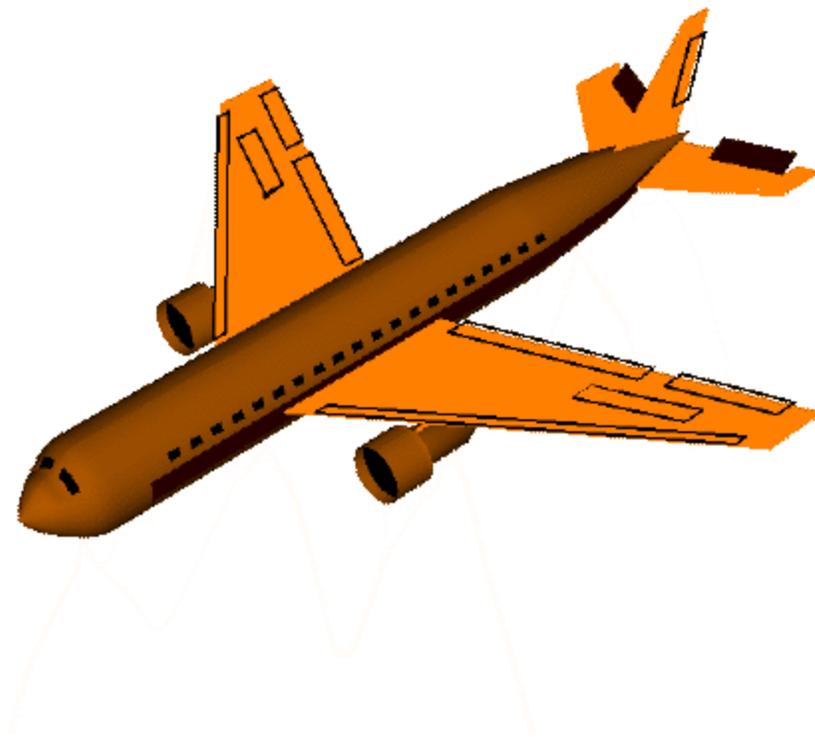
# Main Block Diagram



# Measuring the Roll of the Plane



# Measuring the Pitch of the Plane



# ADXRS300 - Angular Rate Sensor

- Contains an internal Gyroscope
- Output voltage proportional to the angular rate about the axis perpendicular to the surface of the chip
- Range of rate: +/- 300 °/sec
- Zero movement: outputs 2.5 V

# Getting an Angle from Angular Rate

- $\text{AngleRate} = K * (\text{ADCVoltage} - \text{ZeroVoltage})$
- K is some constant (Degs/sec/volt)
- $\text{Angle} = \text{Angle} + \text{AngleRate} * \text{deltaT}$
- May need calibration for ZeroVoltage

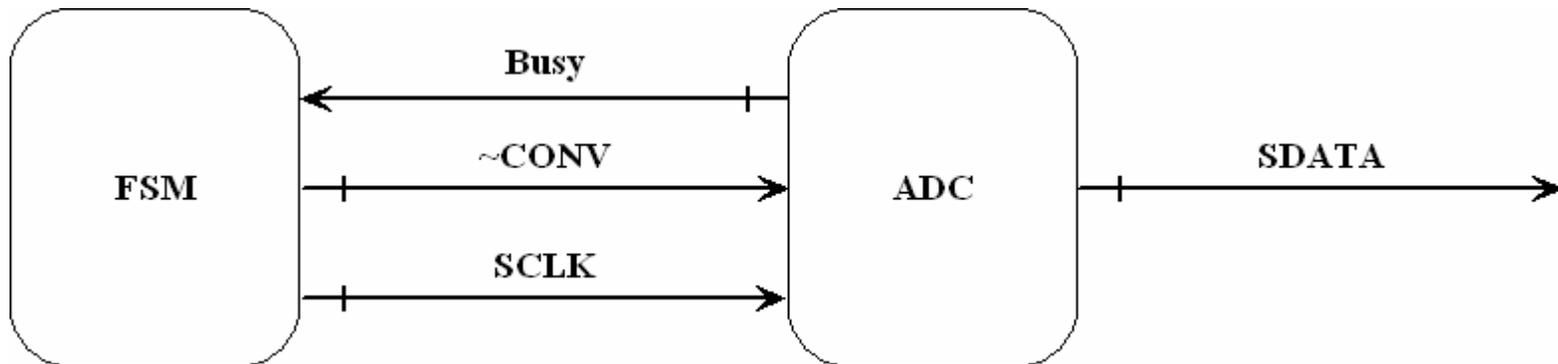
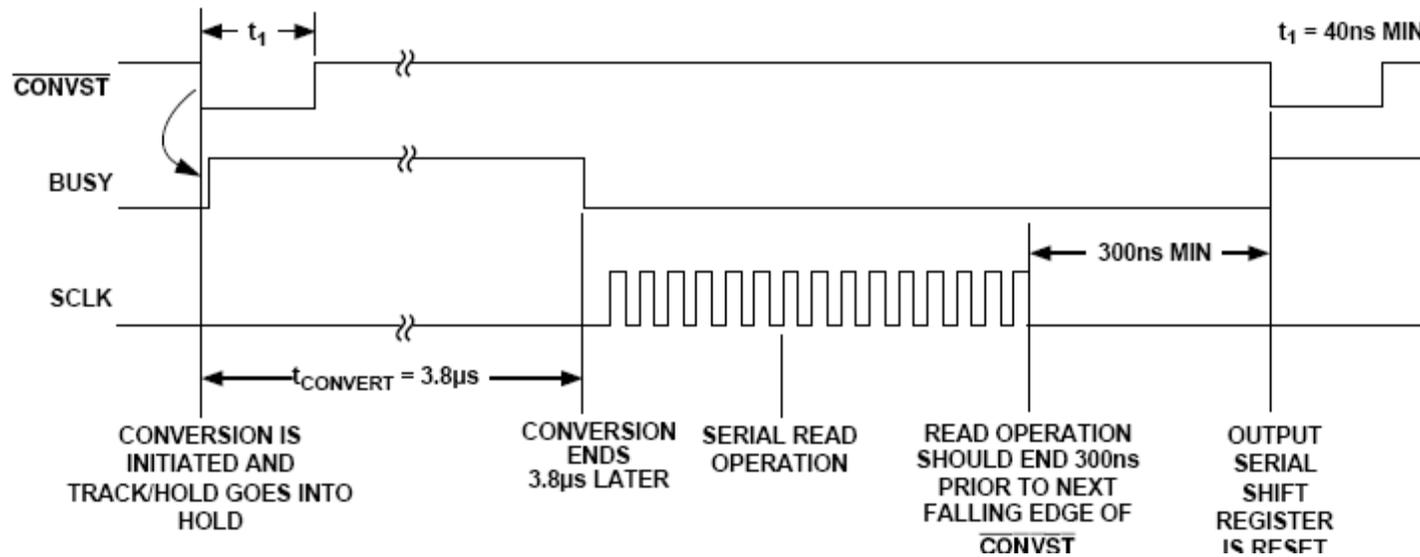
# Interfacing the ADXRS300

- Will use an analog to digital converter AD7895AN-2
- Output of the AD7895 is 12 bits
- Uses a reference potential of 2.5 volts
- Serial Output

# Interfacing the ADXRS300

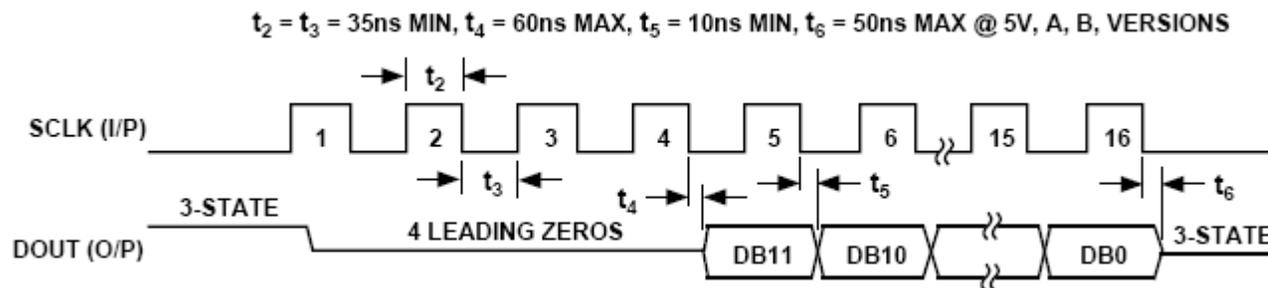
- Bandwidth of the ADXRS300: 400Hz
- Minimum sampling rate for ADC is 800Hz
- We'll use 10 KHz sampling rate

# Timing Operation Diagram



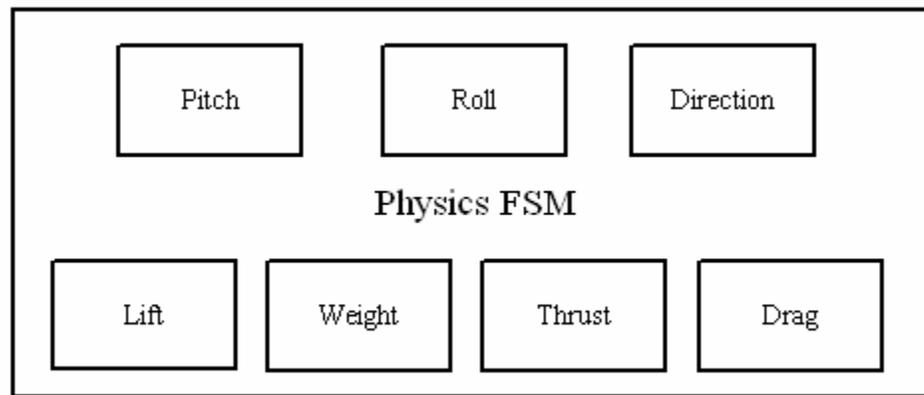
# Data Read Operation

- AD7895 uses 16 clock cycles to output the digital data bits resulting from the conversion
- It outputs 4 leading zeros, then the 12 bits of actual data, starting with the

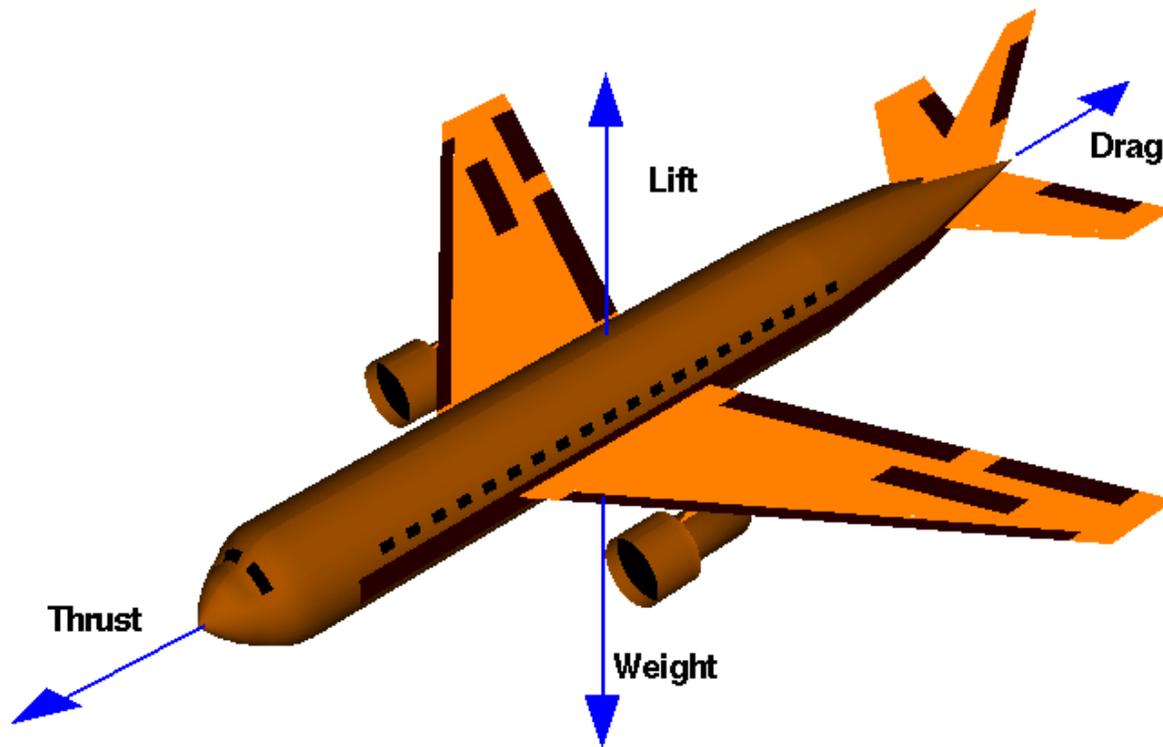


# Forces Determined in Physics Module

- Forces and Angular Velocities determined in Minor FSM
- Positions and Angles calculated in Physics FSM



# Forces on an Airplane

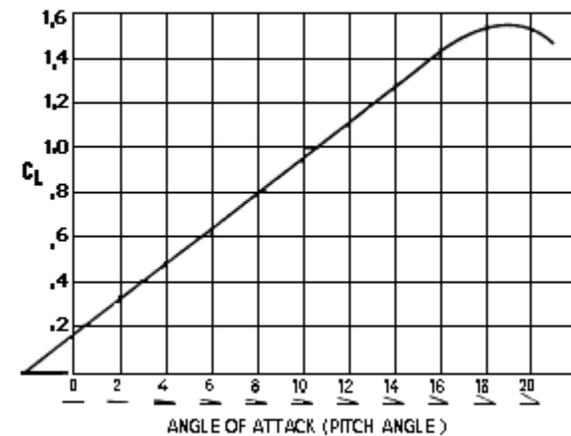


# Force equations

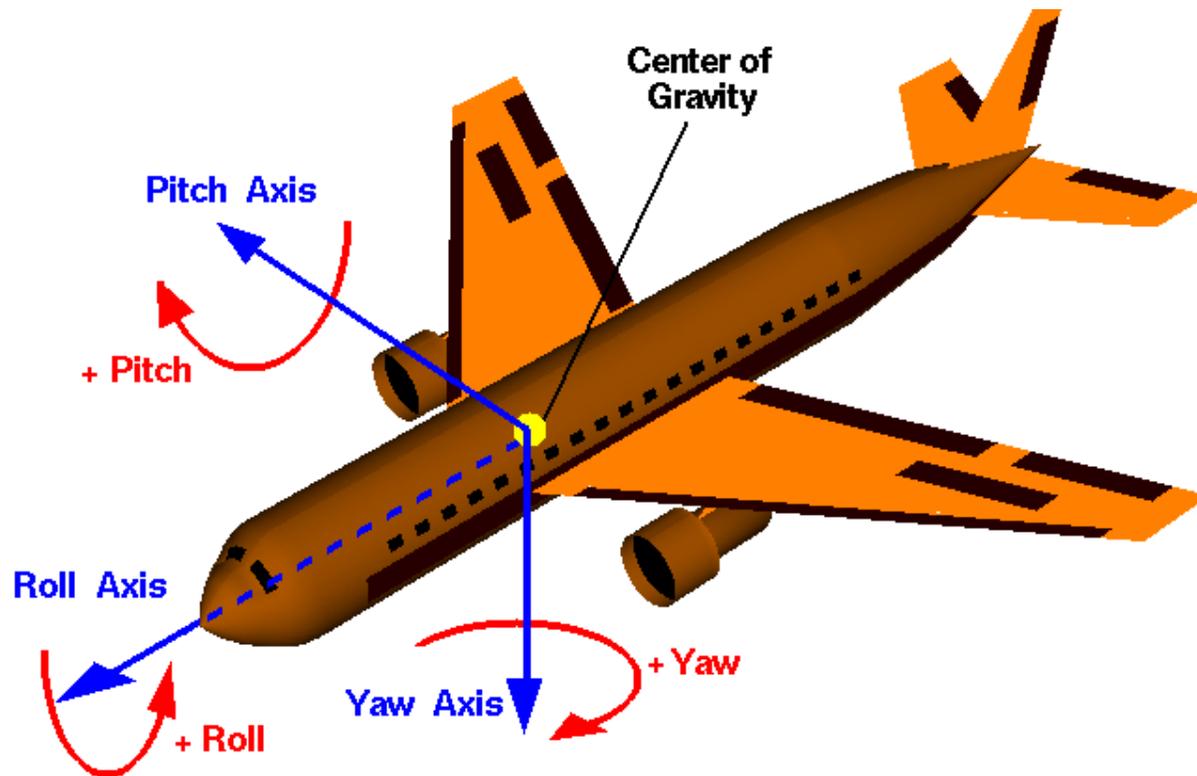
- Thrust:  $F = ma$
- Weight:  $F = mg$

$$\text{lift} = C_L \times \left(\frac{1}{2} \rho V^2\right) \times S$$

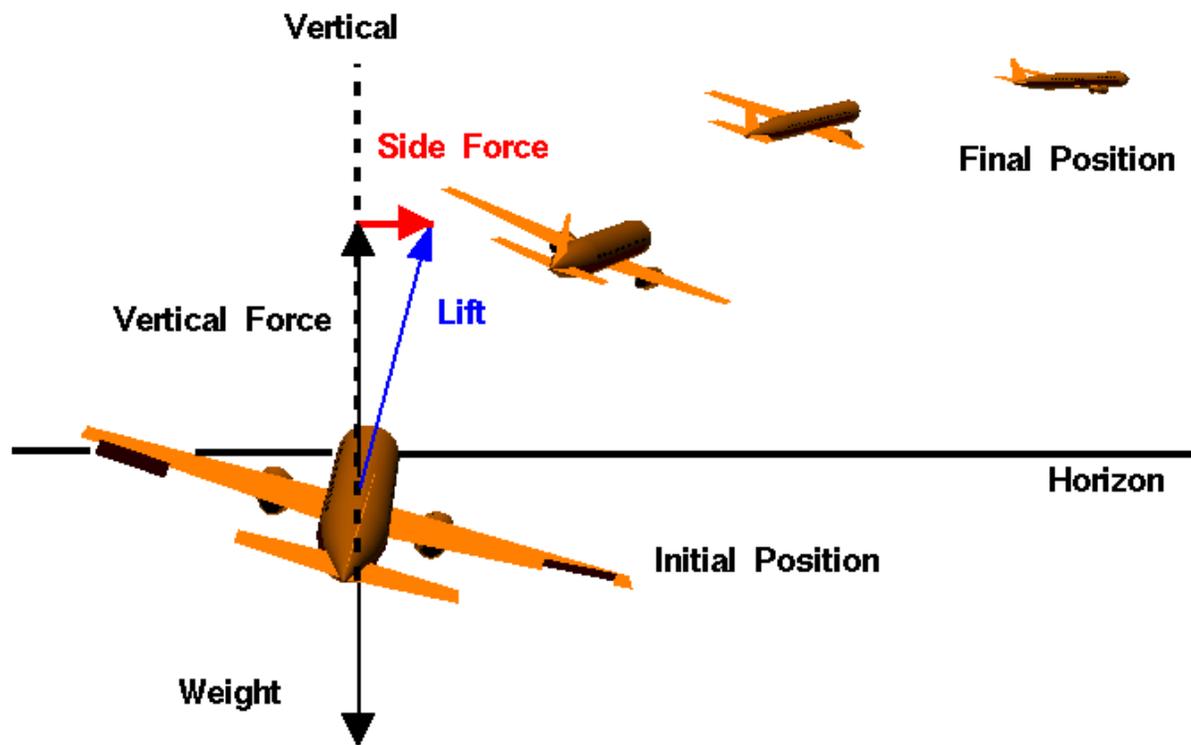
$$\text{drag} = C_D \times \left(\frac{1}{2} \rho V^2\right) \times A$$



# Aircraft Rotations



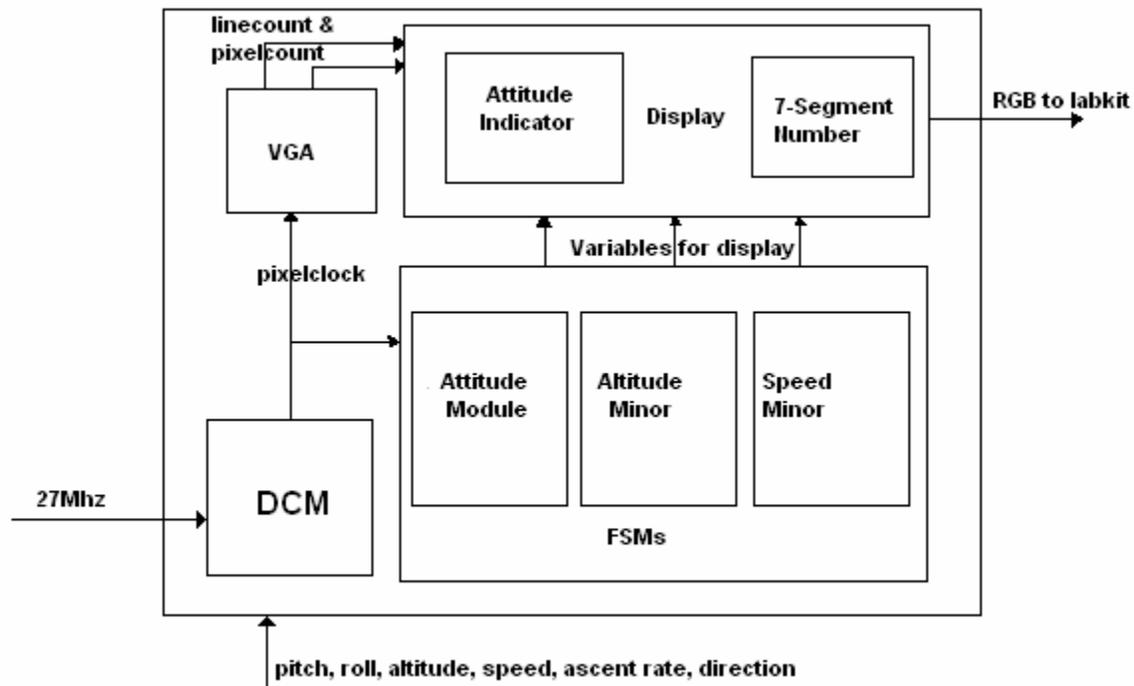
# Rotation produces Vectors



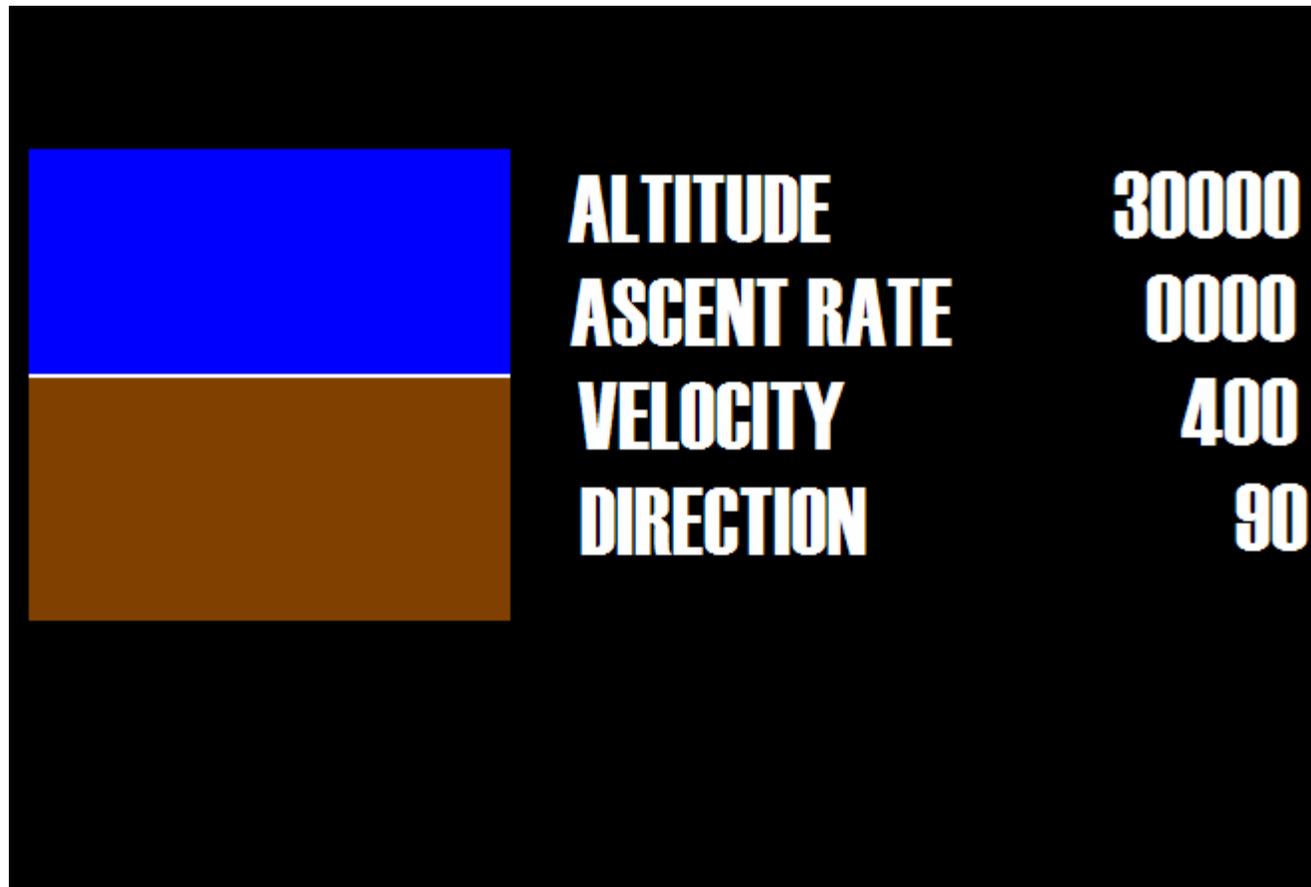
# Displaying the State of the Flight

- The pilot flying the plane stands in front of a monitor that displays the main features of an airplane console, including an attitude indicator and a display for altitude, ascent rate, and velocity.

# Video Display Block Diagram



# Screenshot



The screenshot shows a data display on a black background. On the left side, there is a vertical bar divided into two colored sections: a blue section on top and a brown section on the bottom. To the right of this bar, four lines of white text display numerical data.

<b>ALTITUDE</b>	<b>30000</b>
<b>ASCENT RATE</b>	<b>0000</b>
<b>VELOCITY</b>	<b>400</b>
<b>DIRECTION</b>	<b>90</b>

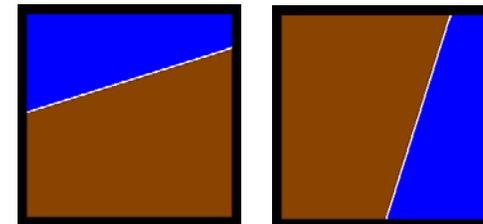
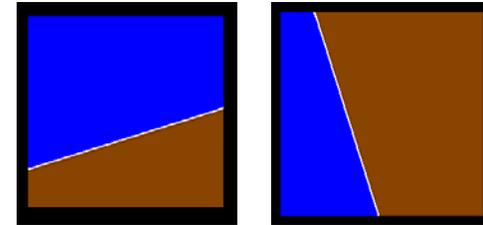
# Displaying numbers

- Approach 1- Instantiate rectangles to form numbers (similar to how MIT logo was made in the Pong game)
- Approach 2- Create and store table of ASCII characters in memory and render characters when they are needed

<b>ALTITUDE</b>	<b>30000</b>
<b>ASCENT RATE</b>	<b>0000</b>

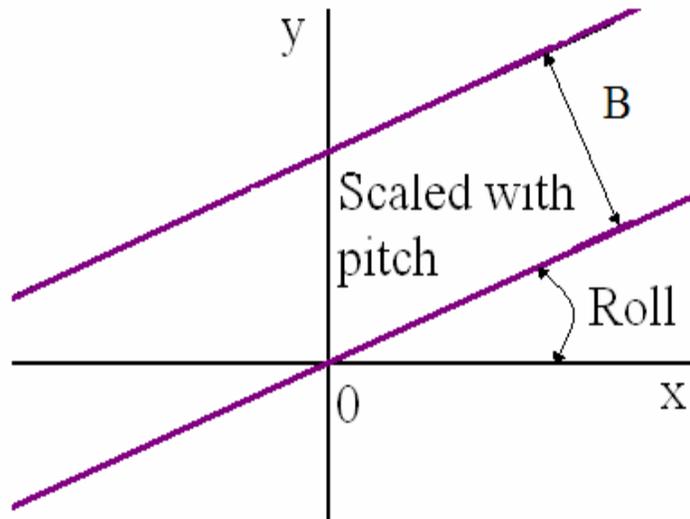
# Attitude Indicator

- The Attitude Indicator Module takes in two angles (pitch and roll).
- The roll of the airplane determines the slope of the white line (horizon).
- The area above is colored blue (sky).
- The area below is colored brown (earth).
- The pitch determines the position of the horizon.



# Attitude Indicator – Algorithm

- The goal is to make the horizon shift and rotate in response to pitch and roll.
- When airplane is flying “sideways,” a different equation is used to draw the line representing the horizon.



When Roll is not  $\pi/2$  or  $-\pi/2$   
$$y = (x + B * \sin(\text{Roll})) * \tan(\text{Roll}) + B * \cos(\text{Roll})$$

When Roll is  $\pi/2$  or  $-\pi/2$   
$$x = -B * \sin(\text{Roll})$$