Lecture notes in Fluid Dynamics (1.63J/2.01J) by Chiang C. Mei, MIT, Spring, 2007

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March 14, 2007

4.4 Buoyant plume from a steady heat source

[Reference]: Gebhart, et. al. (Jalluria, Maharjan, Saammakia), **Buoyancy-induced Flows** and **Transport**, 1988, Hemisphere Publishing Corporation

Let $\tilde{T} = T - T_{\infty}$ = temperature variation where T_{∞} is a constant (no ambient stratification). For a strong enough heat source, we expect the boundary layer behavior,

$$\frac{\partial}{\partial r} \gg \frac{\partial}{\partial x}, \ u \gg v, \ \frac{\partial p}{\partial r} \cong 0$$

The boundary layer equations are

$$\frac{\partial(ru)}{\partial x} + \frac{\partial(rv)}{\partial r} = 0 \tag{4.4.1}$$

$$u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial r} = g\beta(T - T_{\infty}) + \frac{\nu}{r}\frac{\partial}{\partial r}\left(r\frac{\partial u}{\partial r}\right)$$
(4.4.2)

$$u\frac{\partial\tilde{T}}{\partial x} + v\frac{\partial\tilde{T}}{\partial r} = \frac{k}{r}\frac{\partial}{\partial r}\left(r\frac{\partial\tilde{T}}{\partial r}\right)$$
(4.4.3)

The centerline r = 0 is an axis of symmetry,

$$v = \frac{\partial u}{\partial r} = \frac{\partial \tilde{T}}{\partial r} = 0 \tag{4.4.4}$$

Far outside the plume $r \to \infty$

$$u \to 0 \text{ and } T \to T_{\infty}, (\tilde{T} \to 0)$$
 (4.4.5)

Rewrite (4.4.3) as

$$\frac{\partial (r u \tilde{T})}{\partial x} + \frac{\partial (r v \tilde{T})}{\partial r} - \tilde{T} \left(\frac{\partial (r u)}{\partial x} + \frac{\partial (r v)}{\partial r} \right)$$
$$= \frac{\partial (r u \tilde{T})}{\partial x} + \frac{\partial (r v \tilde{T})}{\partial r} = k \frac{\partial}{\partial r} \left(r \frac{\partial \tilde{T}}{\partial r} \right)$$
(4.4.6)

after using continuity. Now integrating the last equation from r = 0 to $r = \infty$

$$\frac{\partial}{\partial x} \int_0^\infty 2\pi r u \tilde{T} dr + 2\pi \int_0^\infty \frac{\partial (r v \tilde{T})}{\partial r} dr$$
$$= k 2\pi \int_0^\infty dr \frac{\partial}{\partial r} \left(r \frac{\partial \tilde{T}}{\partial r} \right)$$

therefore

$$\frac{\partial}{\partial x} \int_0^\infty 2\pi r u \tilde{T} dr + 2\pi r v \tilde{T} \Big|_0^\infty = 2\pi k \left(r \frac{\partial \tilde{T}}{\partial r} \right)_{r=0}^{r=\infty}$$
(4.4.7)

Using the boundary conditions, we get or

$$\int_0^\infty 2\pi r u \tilde{T} dr = \text{constant}$$

Note that

$$\int_0^\infty 2\pi r dr \, u\rho C\tilde{T} = \text{rate of buoyancy flux}$$

= rate of heat flux
= $Q(\text{given rate of heat release at } x = 0)$

therefore,

$$Q = \int_0^\infty 2\pi r dr \rho u C \tilde{T} \tag{4.4.8}$$

This is a boundary condition.

Let the stream function ψ be defined by

$$ru = \frac{\partial \psi}{\partial r}, \qquad rv = -\frac{\partial \psi}{\partial x}$$
 (4.4.9)

(4.4.1) is automatically satisfied. From the momentum equation:

$$\left(\frac{1}{r}\frac{\partial\psi}{\partial r}\right)\frac{1}{r}\frac{\partial^{2}\psi}{\partial x\partial r} - \frac{1}{r}\frac{\partial\psi}{\partial x}\frac{\partial}{\partial r}\left(\frac{1}{r}\frac{\partial\psi}{\partial r}\right) = g\beta\tilde{T} + \frac{\nu}{r}\frac{\partial}{\partial r}\left[r\frac{\partial}{\partial r}\left(\frac{1}{r}\frac{\partial\psi}{\partial r}\right)\right]$$
(4.4.10)

From the energy equation

$$\frac{1}{r}\frac{\partial\psi}{\partial r}\frac{\partial\tilde{T}}{\partial x} - \frac{1}{r}\frac{\partial\psi}{\partial x}\frac{\partial\tilde{T}}{\partial r} = \frac{k}{r}\frac{\partial}{\partial r}\left(r\frac{\partial\tilde{T}}{\partial r}\right)$$
(4.4.11)

and from the buoyancy flux condition

$$Q = 2\pi\rho C \int_0^\infty r dr \left(\frac{1}{r}\frac{\partial\psi}{\partial r}\right) \tilde{T}$$
(4.4.12)

Try a similarity solution with the one-parameter transformation

$$x - \lambda^a x^*, \quad r = \lambda^b r^*, \quad \psi = \lambda^c \psi^*, \quad \tilde{T} = \lambda^d T^*$$

From (4.4.10),

$$\lambda^{2c-4b-a} = \lambda^{2c-4b-a} = \lambda^d = \lambda^{c-4b} \tag{4.4.13}$$

from (4.4.11)

$$\lambda^{c+d-2b-a} = \lambda^{d-2b} \tag{4.4.14}$$

and from (4.4.12)

$$\lambda^{c+d} = 1 \tag{4.4.15}$$

From these three equations we get

$$\frac{c}{a} = 1, \quad \frac{b}{a} = \frac{1}{2}, \quad \frac{d}{a} = -1$$

We leave it as an exercise to show that the similarity variable can be taken to be

$$\eta = \frac{r}{x^{1/2}} \tag{4.4.16}$$

and the similarity solutions to be

$$\psi = xF(\eta), \text{ and } \tilde{T} = x^{-1}G(\eta)$$
 (4.4.17)

After much algebra, and noting

$$\frac{\partial \eta}{\partial r} = \frac{1}{x^{1/2}}, \quad \frac{\partial \eta}{\partial x} = -\frac{1}{2}\frac{r}{x^{3/2}} = -\frac{1}{2}\frac{r}{x^{1/2}}\frac{1}{x} = -\frac{\eta}{2x}$$

we get from (4.4.10)

$$\nu F''' + \left(\frac{F'}{\eta}\right)' (F - \nu) + g\beta\eta G = 0$$
(4.4.18)

and from (4.4.11)

$$k(\eta G')' + (FG)' = 0 \tag{4.4.19}$$

Before integrating, let us normalize :

$$\eta = \alpha \bar{\eta}, \quad F = \gamma \bar{F}, \quad G = \sigma \bar{G}.$$
 (4.4.20)

It follows from (4.4.18) that

$$\frac{\nu\gamma}{\alpha^3}\bar{F}''' + \frac{\gamma}{\alpha^3}\left(\frac{\bar{F}'}{\bar{\eta}}\right)'(\gamma\bar{F}-\nu) + g\beta\alpha\sigma\bar{\eta}\bar{G} = 0$$
(4.4.21)

where prime denotes $d/d\bar{\eta}$. Setting $\gamma = \nu$ and

$$\frac{\nu^2}{\alpha^3} = g\beta\alpha\sigma$$

which relates σ and α ,

$$\sigma = \frac{\nu^2}{g\beta\alpha^4} \tag{4.4.22}$$

we get

$$\bar{F}''' + \left(\frac{\bar{F}'}{\bar{\eta}}\right)' (\bar{F} - 1) + \bar{\eta}\bar{G} = 0$$
(4.4.23)

Similar normalization of (4.4.19) gives

$$\frac{k\alpha\sigma}{\alpha^2}(\bar{\eta}\bar{G}')' + \frac{\gamma\sigma}{\alpha}(\bar{F}\bar{G})' = 0$$
(4.4.24)

. which can be simplified to

$$(\bar{\eta}\bar{G}')' + P_r(\bar{F}\bar{G})' = 0 \tag{4.4.25}$$

. where

$$P_r = \frac{\nu}{k} = \text{Prandtl Number}$$
 (4.4.26)

For water $\nu = 10^{-2} cm^2/s$, $k = 1.42 cm^2/s$, hence Pr = 7. For air $\nu = 0.145 cm^2/s$, $k = 0.202 cm^2/s$, hence Pr = 0.75.

We now integrate (4.4.25) to give

$$\bar{\eta}\bar{G}' + P_r\bar{F}\bar{G} = \text{constant}$$

Since $\psi(x,0) = 0$, we must have $\overline{F}(0) = 0$; the constant above is zero.

$$\bar{\eta}\bar{G}' + P_r\bar{F}\bar{G} = 0 \tag{4.4.27}$$

Equation (4.4.27) can be written

$$\frac{\bar{G}'}{\bar{G}} = -P_r \frac{\bar{F}}{\bar{\eta}}, \text{ or } \frac{d \ln \bar{G}}{d\bar{\eta}} = -P_r \frac{\bar{F}}{\bar{\eta}}$$

$$\ln \bar{G} = -P_r \int_0^{\bar{\eta}} \frac{\bar{F}}{\bar{\eta}} d\bar{\eta} + \text{constant}$$

$$\bar{G}(\bar{\eta}) = \bar{G}(0) \exp\left(-P_r \int_0^{\bar{\eta}} \frac{\bar{F}}{\bar{\eta}} d\bar{\eta}\right)$$
(4.4.28)

Substituting Eqn. (4.4.28) into Eqn. (4.4.23), the resulting equation for \overline{F} must be integrated numerically.

Now let us find the boundary conditions for F or \overline{F} .

Eqn. (4.4.8) becomes

$$\frac{Q}{2\pi\rho C} = \int_0^\infty dr \, r \left(\frac{1}{r}\frac{\partial\psi}{\partial r}\right) \frac{G(\eta)}{x} = \int_0^\infty dr \frac{r}{r} x^{1/2} F' \frac{G}{x} = \int_0^\infty d\eta (F'G) = \nu\sigma \int_0^\infty d\bar{\eta} (\bar{F}'\bar{G})$$
(4.4.29)

Therefore,

$$\int_0^\infty d\bar{\eta} \,\bar{F}'\bar{G} = \frac{Q}{2\pi\rho C\nu\sigma} \tag{4.4.30}$$

Let us choose

$$\frac{Q}{2\pi\rho C\nu\sigma} = 1 \tag{4.4.31}$$

so that

$$\int_0^\infty d\bar{\eta} \,\bar{F}'\bar{G} = 1 \tag{4.4.32}$$

is the boundary condition for \overline{F} and \overline{G} . Now (4.4.31) defines σ , the scale of G. Note that larger Q implies larger σ and smaller α . Thus a stronger heat source leads to a greater centerline temperature and a thinner plume. Also,

$$u \to 0$$
 as $r \to \infty$

hence

$$u = \frac{1}{r}\psi_r = \frac{F'}{\eta} = \frac{\nu}{\alpha^2}\frac{\bar{F}'}{\bar{\eta}} \to 0, \text{ as } \eta \sim \bar{\eta} \to \infty$$

The radial velocity is, in general

$$v = \frac{1}{r}\psi_x = \frac{1}{r}\left(F - \eta\frac{F'}{2}\right)$$

Since

 $v \to 0$ as $\eta \to 0$,

we must have,

F(0) = 0.

Clearly

 $\bar{F}(\bar{\eta}) = 0 \text{ as } \bar{\eta} \to 0$ (4.4.33)

The numerical results by Mollendorf & Gelhart, 1974, are shown in Figs. 4.4.1, for various Prandtl numbers. A schlierian photograph due to Gebhart (copied from Van Dyke **An Album of Fluid Motion**) is hown in Figure fig:plumeVD.

<u>Remark</u>:

$$u = \frac{1}{r} \frac{\partial \psi}{\partial r} = \frac{F'}{\eta} \left(= \frac{x}{r} F' \frac{1}{x^{1/2}} \right)$$

Along the centerline $u(x,0) = \left(\frac{F'}{\eta}\right)_0$ = constant depending on P_r . Why? Buoyancy acceleration is counteracted by entrainment.

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<u>Remark</u>: Let the radius of the plume be a which varies as

$$a \sim x^{1/2}$$

This is consistent with the behavior that $u \sim x^0$, and $\tilde{T} \sim x^{-1}$, since

$$a^2 u \tilde{T} = Q$$

On the other hand the mass flux rate is

$$ua^2 \sim x$$

and the momentum flux rate is

$$u^2 a^2 \sim x$$

hence both approach zero at the source. Thus a plume is the result of energy source, not of mass or momentum.



Figure 4.4.1 Velocity profiles in an axisymmetric plume. (From Mollendorf and Gebhart, 1974.)

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1.0 Pr = 0.7 G 2.0 0.5 7.0 0.0 L 7 10 5 6 8 9 2 3 4 7 1 η

Figure 4.4.2 Temperature profiles in an axisymmetric plume. (From Mollendorf and Gebhart, 1974.)



Figure 4.4.2: A 2D thermal plume from a line heat source. From Van Dyke, photo by Gebhart, Pera and Schoor 1970,